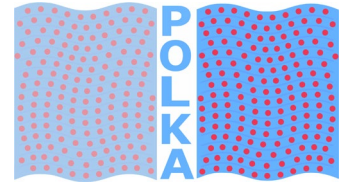


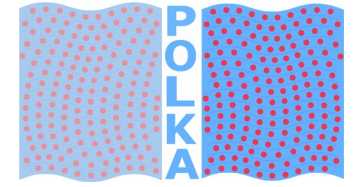
Introduction to plane wave decomposition and two-port analysis in flow duct acoustics

HANS BODÉN

CONTENTS



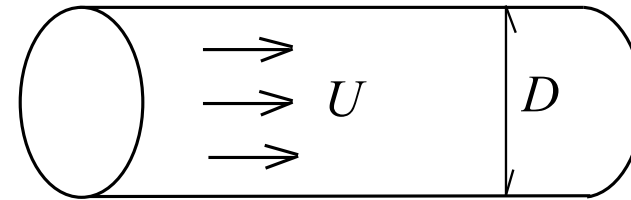
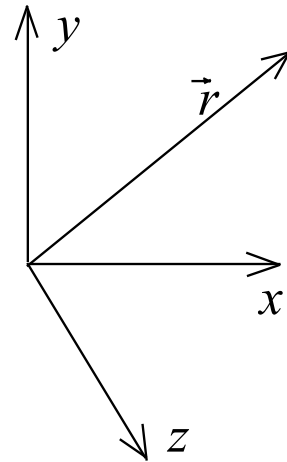
- Introduction.
- Passive one-port measurement techniques - The two-microphone technique.
Sources of errors.
- Passive two-port measurement techniques.
Flow noise suppression



Sound Propagation in a Duct

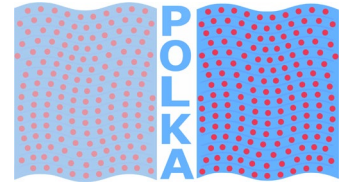
Wave Equation

$$\nabla^2 p - \frac{1}{c^2} \frac{\partial^2 p}{\partial t^2} = 0$$



Modified Wave Equation

$$\nabla^2 p - \frac{1}{c^2} \left(\frac{\partial}{\partial t} + U \frac{\partial}{\partial x} \right)^2 p = 0$$



Propagating Pressure Wave

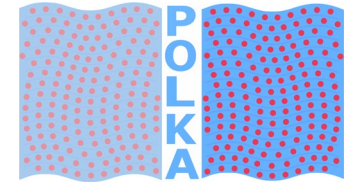
$$\mathbf{p} = \hat{p} e^{i\mathbf{k}_x x} e^{i\omega t}$$

$$\mathbf{k}_x^+ = \frac{k}{1+M}$$

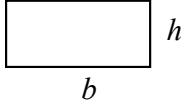

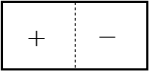



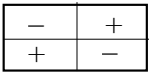



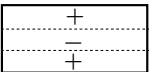

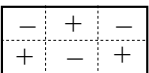

$$\mathbf{k}_x^- = \frac{k}{1-M}$$

$$\mathbf{p} = \hat{p}_+ e^{-i\mathbf{k}_x^+ x} + \hat{p}_- e^{i\mathbf{k}_x^- x}$$

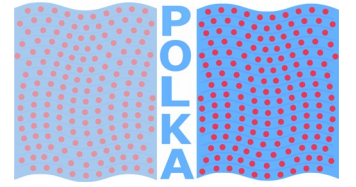
Cut-on of higher order modes



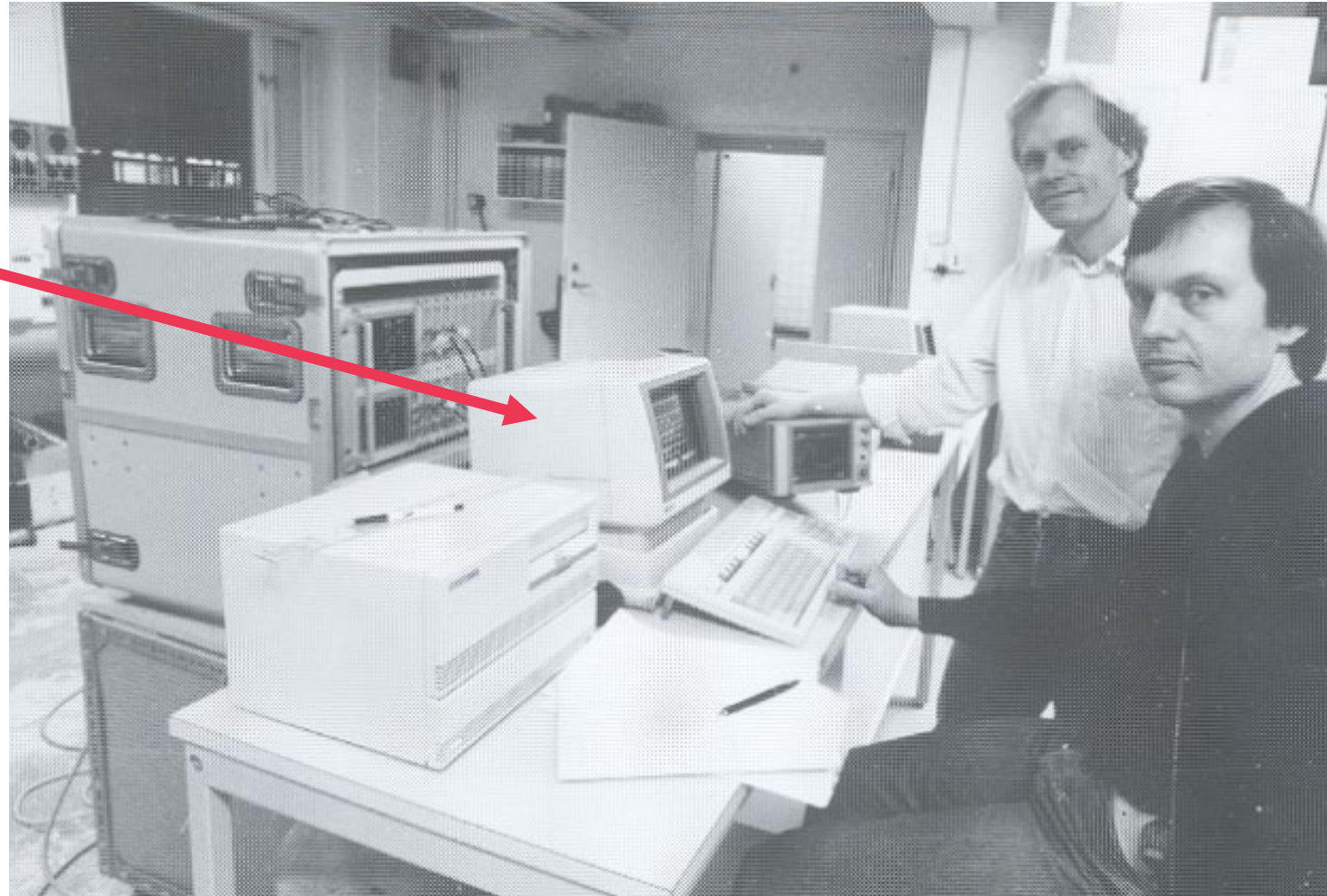
This is the limitation of the plane wave region

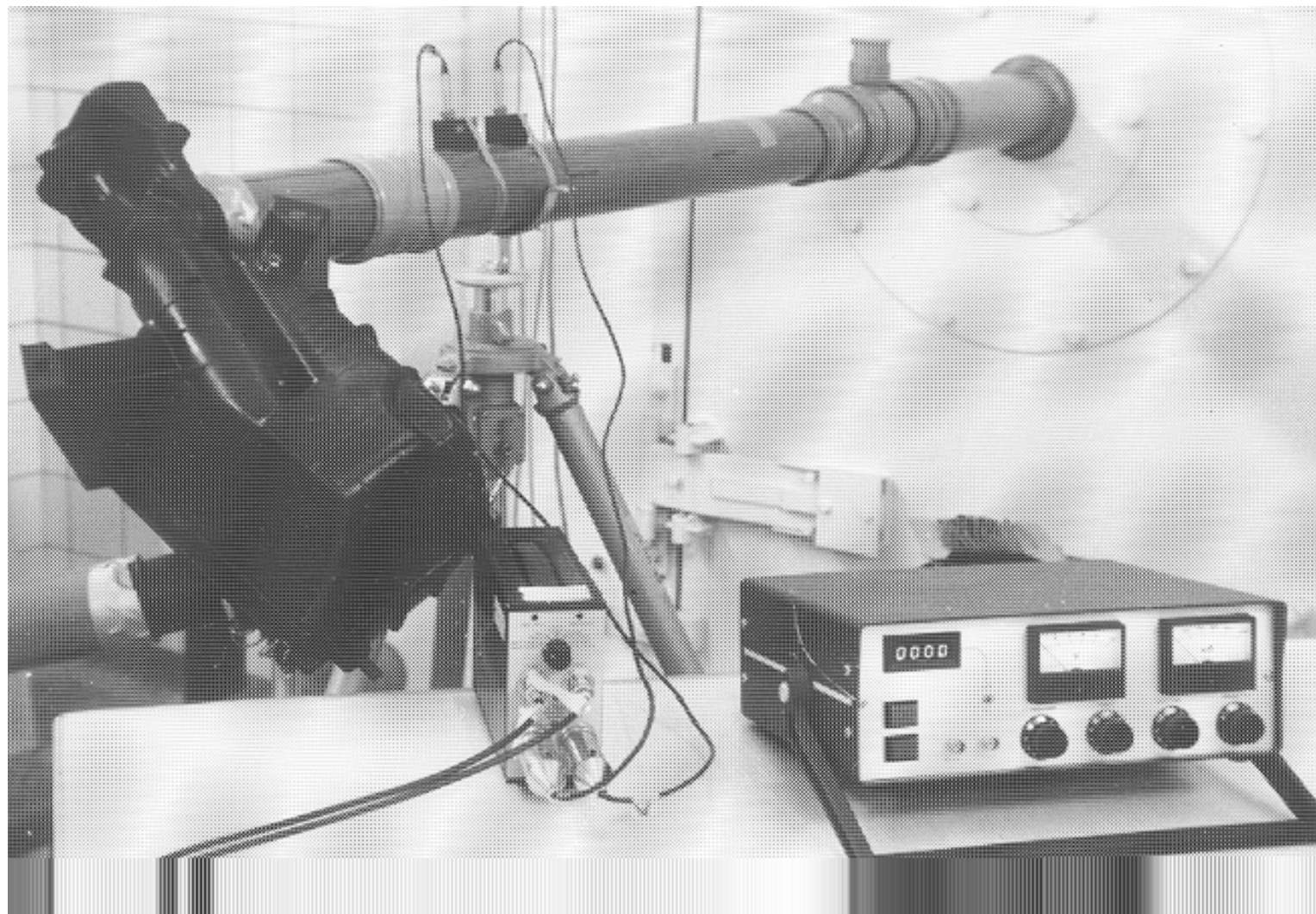
Rectangular cross section		Circular cross section	
$f_{10}^c = c/2b$		$f_{01}^c = 1.841c/\pi D$	
$f_{01}^c = c/2h$		$f_{02}^c = 3.054c/\pi D$	
$f_{11}^c = \frac{c}{2} \left(\frac{1}{b^2} + \frac{1}{h^2} \right)^{1/2}$		$f_{10}^c = 3.832c/\pi D$	
$f_{02}^c = c/b$		$f_{03}^c = 4.201c/\pi D$	
$f_{20}^c = c/h$		$f_{04}^c = 5.318c/\pi D$	
$f_{21}^c = \frac{c}{2} \left(\frac{4}{b^2} + \frac{1}{h^2} \right)^{1/2}$		$f_{11}^c = 5.331c/\pi D$	

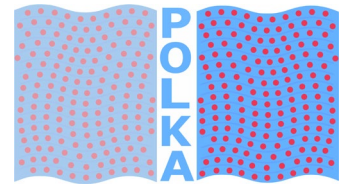
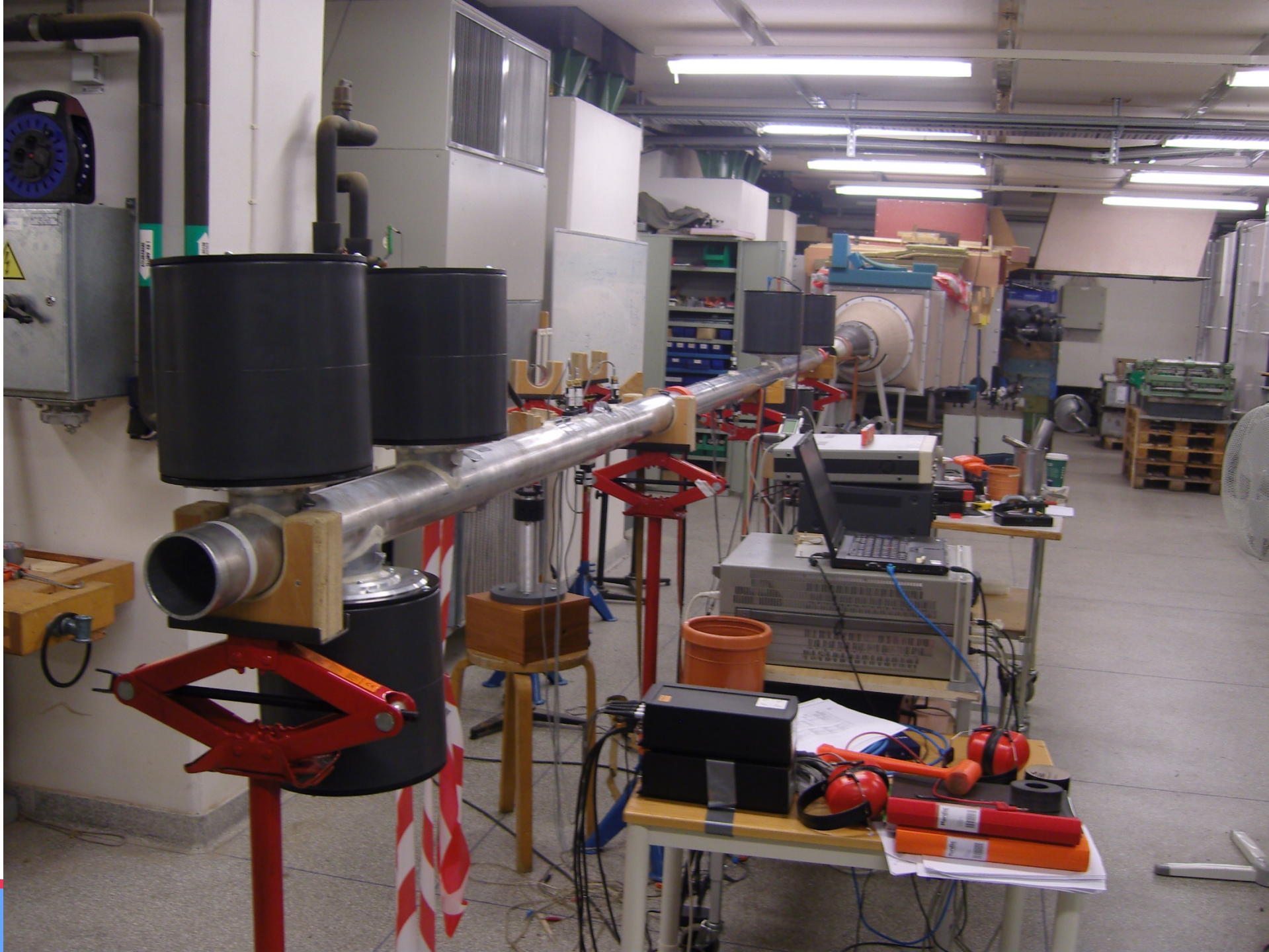
INTRODUCTION

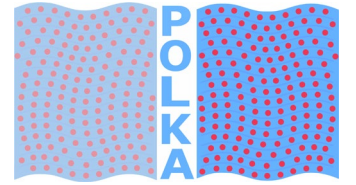


First PC in
the lab









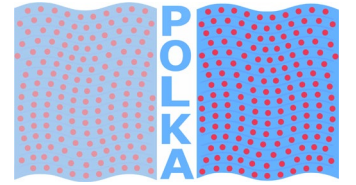
Influence of errors on the two-microphone method for measuring acoustic properties in ducts

Hans Bodén and Mats Åbom

Department of Technical Acoustics, Royal Institute of Technology, 100 44 Stockholm, Sweden

(Received 6 December 1984; accepted for publication 9 September 1985)

Using the two-microphone method, acoustic properties in ducts, as, for example, reflection coefficient and acoustic impedance, can be calculated from a transfer function measurement between two microphones. In this paper, a systematic investigation of the various measurement errors that can occur and their effect on the calculated quantities is made. The input data for the calculations are the measured transfer function, the microphone separation, and the distance between one microphone and the sample. First, errors in the estimate of the transfer function are treated. Conclusions concerning the most favorable measurement configuration to avoid these errors are drawn. Next, the length measurement errors are treated. Measurements were made to study the question of microphone interference. The influence of errors on the calculated quantities has been investigated by numerical simulation. From this, conclusions are drawn on the useful frequency range for a given microphone separation and on the magnitude of errors to expect for different cases.



Error analysis of two-microphone measurements in ducts with flow

Mats Åbom and Hans Bodén

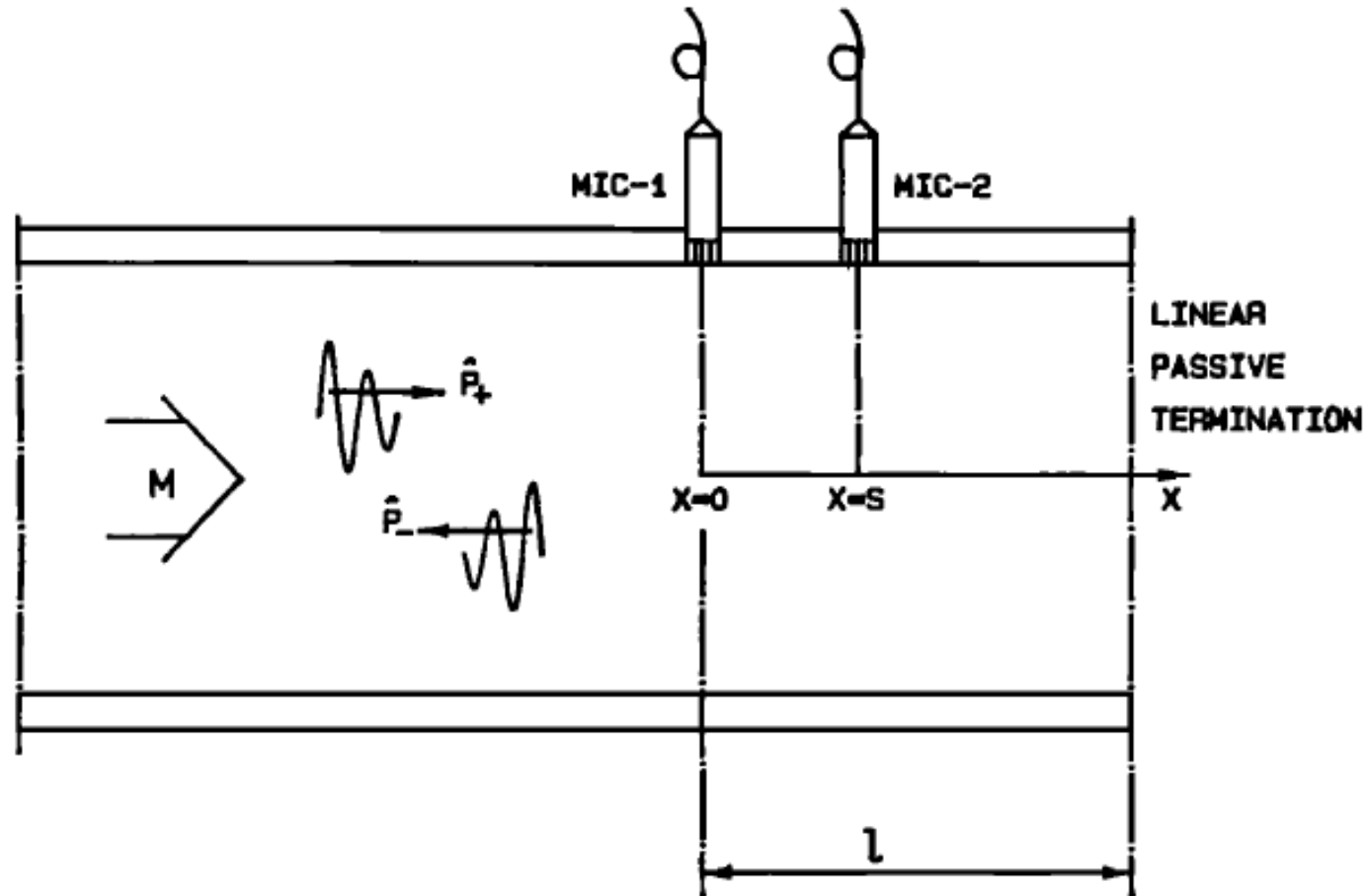
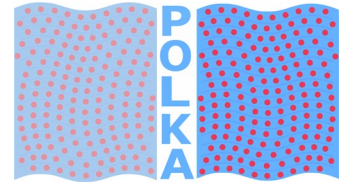
Department of Technical Acoustics, Royal Institute of Technology, S-100 44 Stockholm, Sweden

(Received 2 February 1987; accepted for publication 18 January 1988)

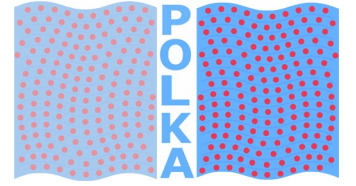
In an earlier work [H. Bodén and M. Åbom, *J. Acoust. Soc. Am.* **79**, 541–549 (1986)] the influence of errors on two-microphone measurements in ducts without flow has been studied. The aim of this article is mainly to extend the earlier work to include the effects of mean flow and also of attenuation during the sound propagation. First, a short review of the various existing two-microphone methods is made. The errors in the measured input data are then analyzed and special attention is paid to the effects of neglected attenuation, nonideal microphones, and flow noise. The influence of errors on the calculated quantities has been investigated and the conclusions from the earlier work have been extended to the case with flow. It is also shown that the neglect of attenuation between the microphones leads to a low-frequency limit for the applicability of the two-microphone method. Finally, a new technique for measuring the Mach number using a two-microphone method is suggested.

Passive one-port measurement techniques

The "Two-Microphone Method"



The Two-Microphone Method



$$p_1(f) = p_+(f) + p_-(f)$$

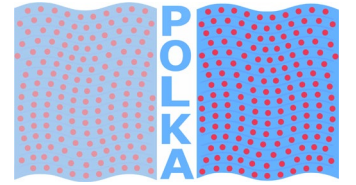
$$p_2(f) = p_+(f) \cdot \exp(-jk_+s) + p_-(f) \cdot \exp(jk_-s)$$

where

$$k_+ = \frac{k}{1+M} \quad k_- = \frac{k}{1-M}$$

A linear system of equations in p_+ and p_- from which the **Reflection Coefficient** at $x=0$ can be calculated

$$R_0(f) = \frac{p_-(f)}{p_+(f)} = \frac{\exp(-jk_+s) - p_2(f)/p_1(f)}{p_2(f)/p_1(f) - \exp(jk_-s)}$$



The Two-Microphone Method

With $H_{12}(f) = \frac{p_2(f)}{p_1(f)}$ we get

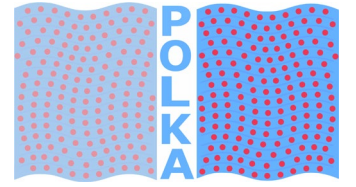
$$R_0(f) = \frac{\exp(-jk_+s) - H_{12}(f)}{H_{12}(f) - \exp(jk_-s)}$$

The reflection coefficient at $x=l$ can be calculated from

$$R_l(f) = R_0(f) \exp\left(\frac{j2kl}{1 - M^2}\right)$$

And the normalised impedance (= one port passive system properties) can be calculated from

$$Z(f) = \frac{p(f)}{\rho_0 c \cdot u(f)} = \frac{1 + R(f)}{1 - R(f)}$$



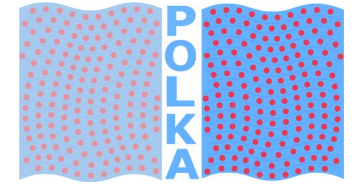
Errors in the Two-Microphone Method

$$R_0(f) = \frac{\exp(-jk_+s) - H_{12}(f)}{H_{12}(f) - \exp(jk_-s)}$$

Errors in the input data: k_+s , k_-s and $H_{12}(f)$

Errors in k_+s and k_-s :

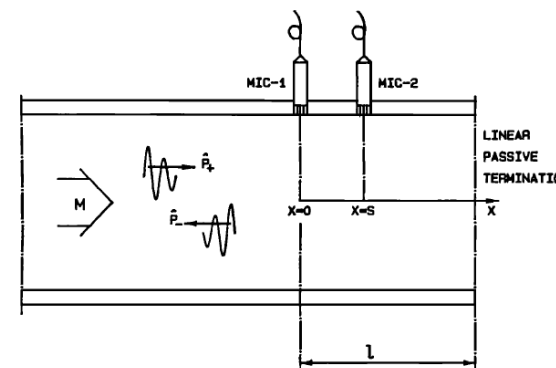
- Uncertainty in determination of k because of mainly turbulent losses
- Uncertainty in Mach-number measurement
- Uncertainty in length measurement: geometric and acoustic length



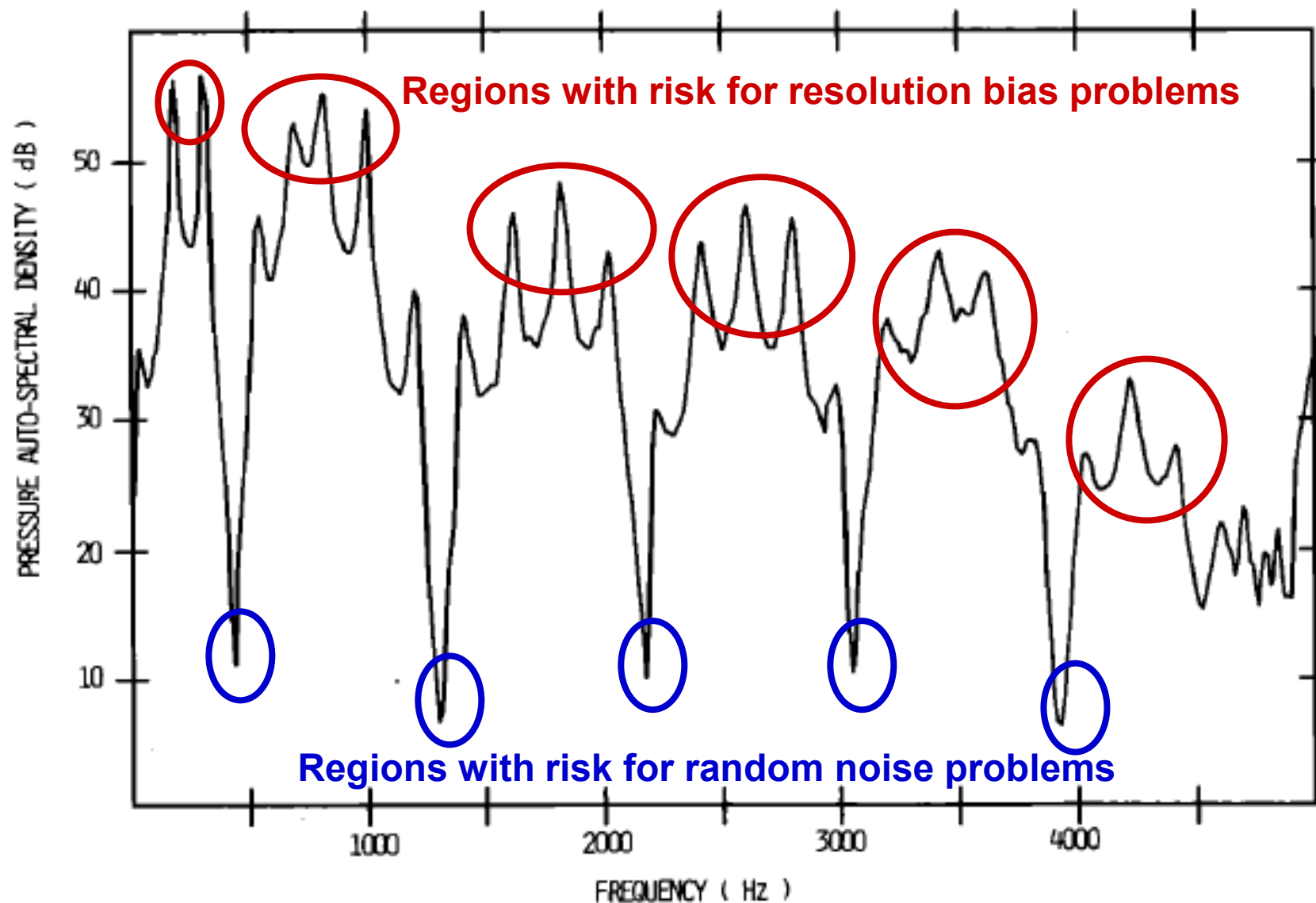
Errors in the Two-Microphone Method

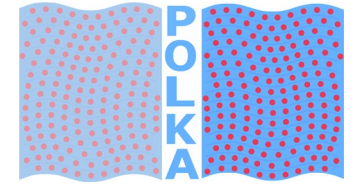
Errors in $H_{12}(f)$

- **Bias errors**, as for instance resolution-bias errors. Problem for long duct systems with many resonances. **Solution – Reflection free terminations.**
- **Random errors** caused by random signals but mainly flow noise disturbances



Pressure autospectral density measured in a duct driven by a loudspeaker and with a rigid termination.

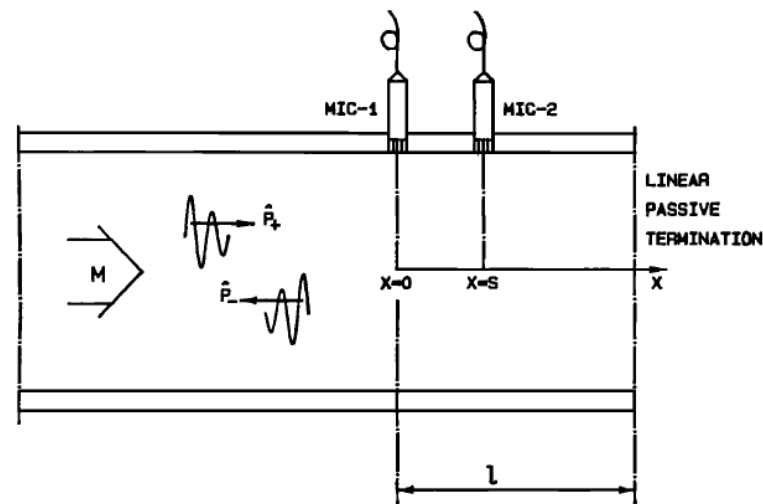




Errors in the Two-Microphone Method

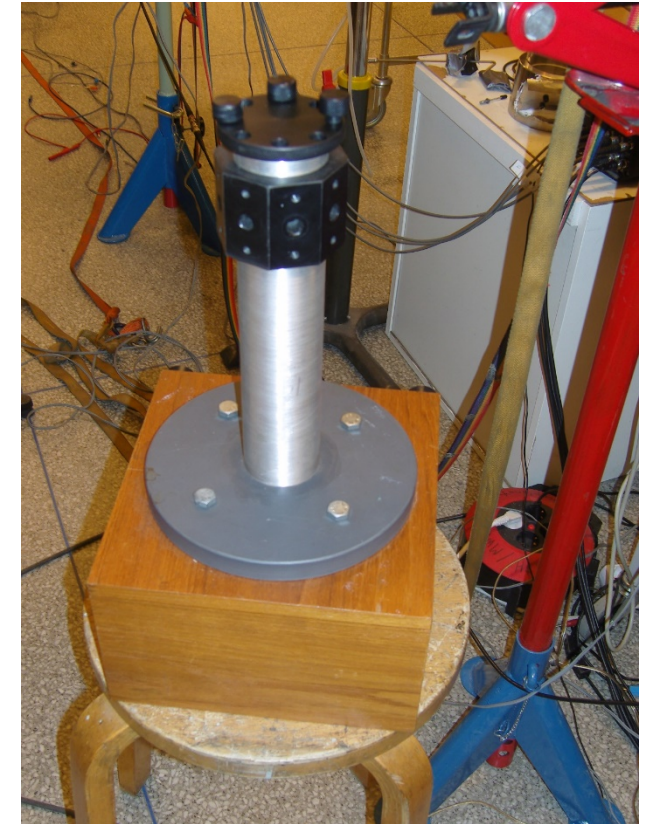
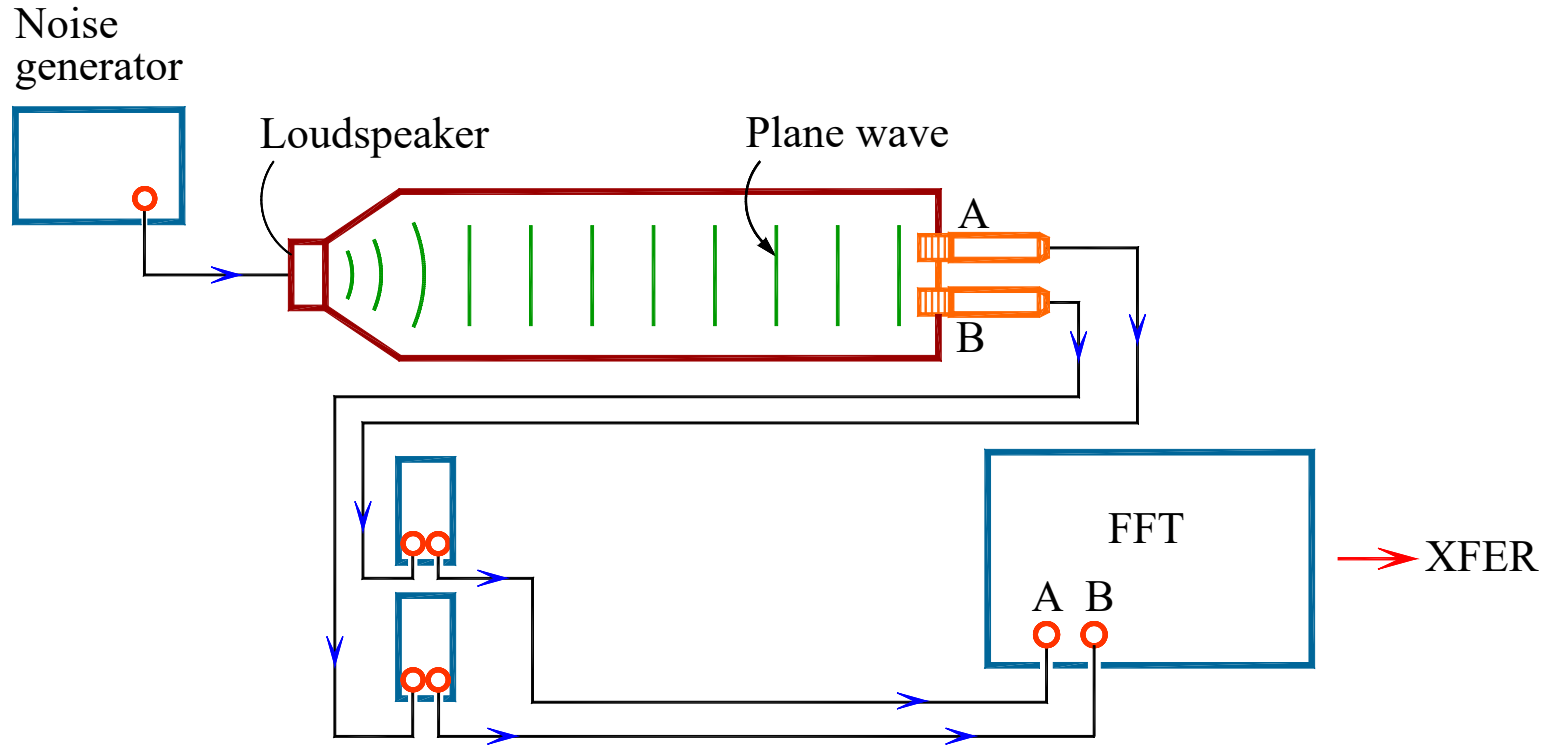
To avoid large sensitivity to the errors in the input data the two-microphone technique should be restricted to the frequency range:

$$0.1 \cdot \pi \cdot (1 - M^2) < ks < 0.8 \cdot \pi \cdot (1 - M^2)$$

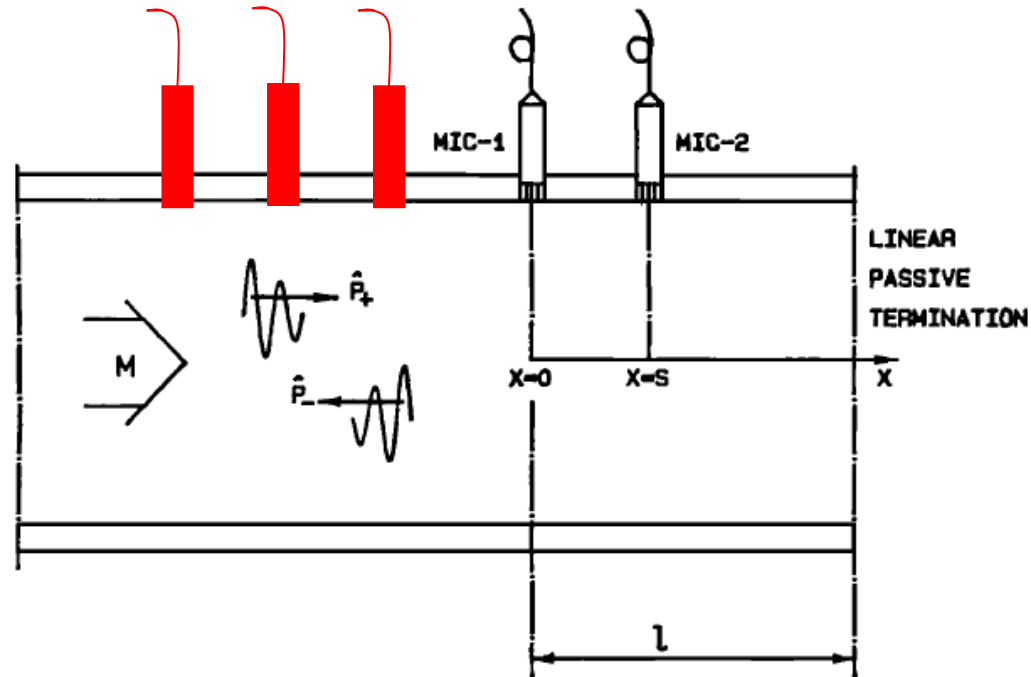


Calibration

Duct method for measurement of $K_{AB}(\omega)$.



Over-determination



Add more
microphones

Can be used to
reduce error

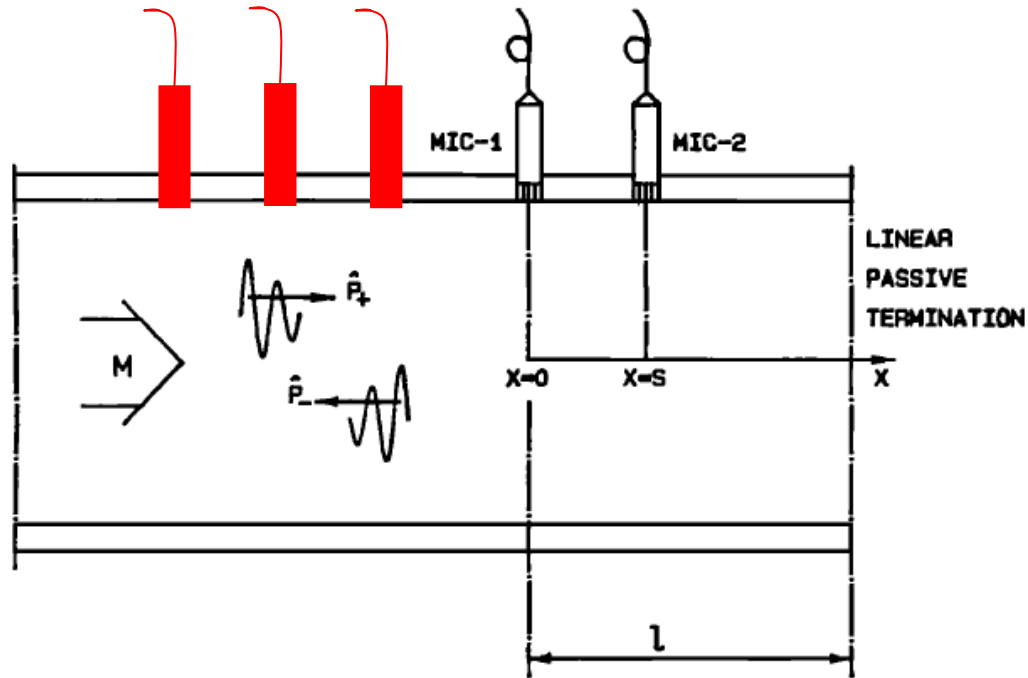
$$p_1(f) = p_+(f) + p_-(f)$$

$$p_2(f) = p_+(f) \cdot \exp(-jk_+s) + p_-(f) \cdot \exp(jk_-s)$$

⋮

$$p_n(f) = p_+(f) \cdot \exp(-jk_+s_n) + p_-(f) \cdot \exp(jk_-s_n)$$

Over-determination



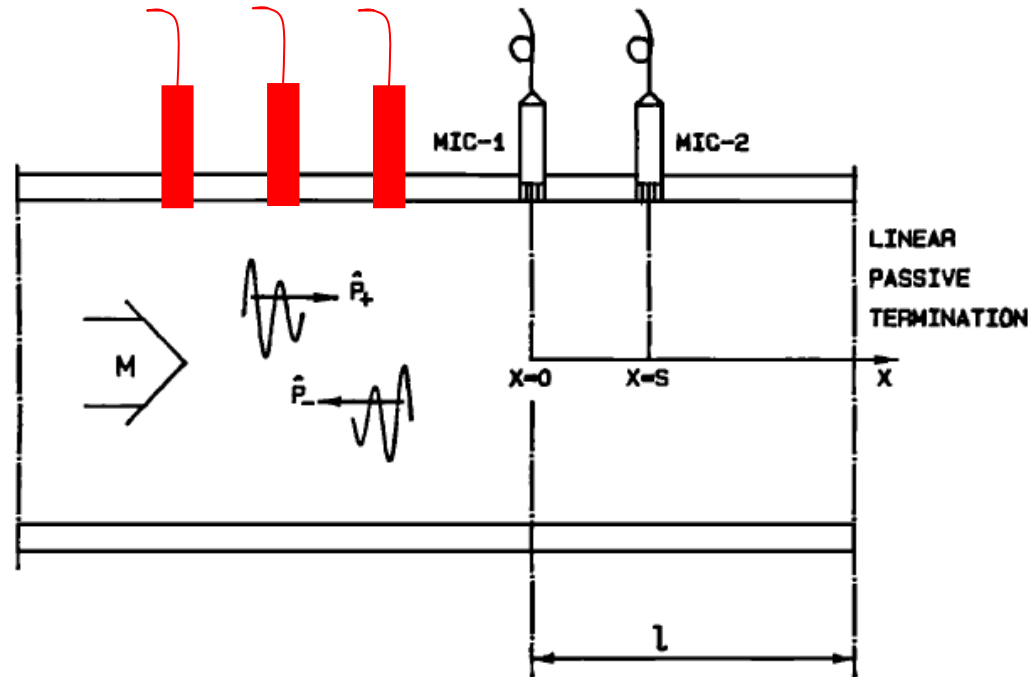
Add more
microphones

Can be used to
reduce error

Accurate experimental two-port analysis of flow generated sound

Andreas Holmberg, Mats Åbom and HansBodén,
Journal of Sound and Vibration 330 (2011) 6336–6354

Treat k_+s and k_-s as unknowns



Add more
microphones

Solve nonlinear
system of
equations for
 k_+s and k_-s

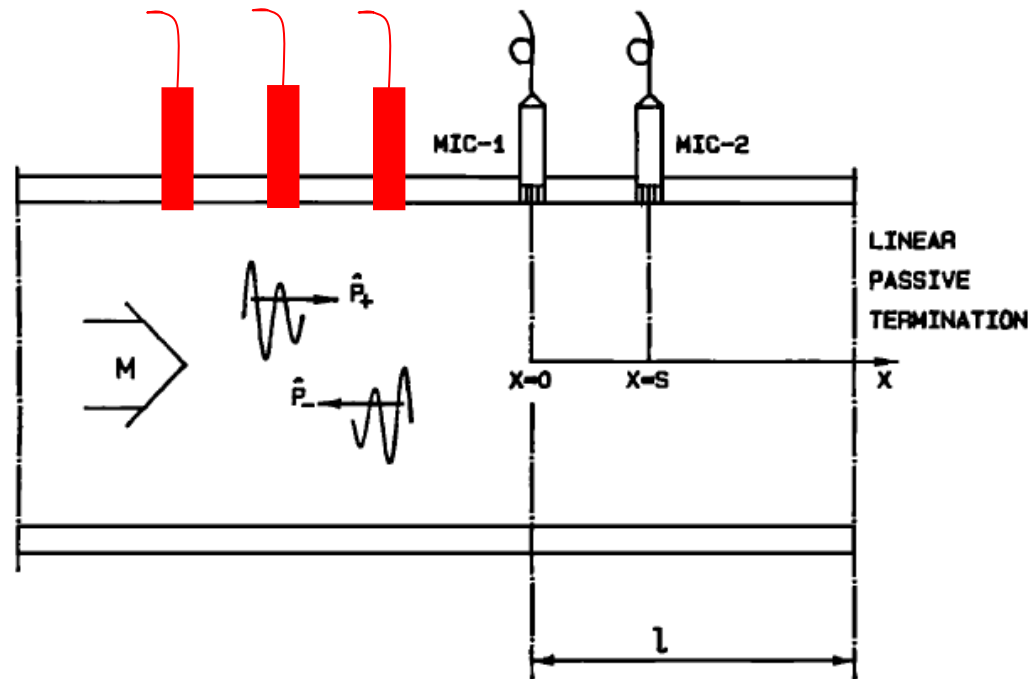
$$p_1(f) = p_+(f) + p_-(f)$$

$$p_2(f) = p_+(f) \cdot \exp(-jk_+s) + p_-(f) \cdot \exp(jk_-s)$$

⋮

$$p_n(f) = p_+(f) \cdot \exp(-jk_+s_n) + p_-(f) \cdot \exp(jk_-s_n)$$

Treat k_+ s and k_- s as unknowns

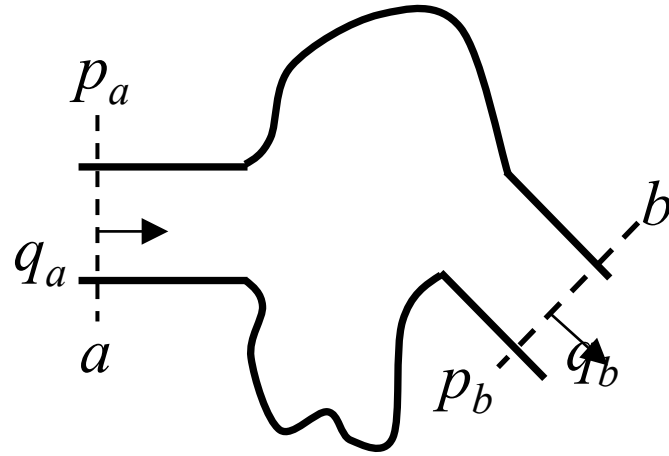
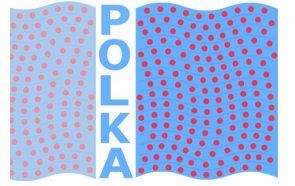


Add at least 4 microphones

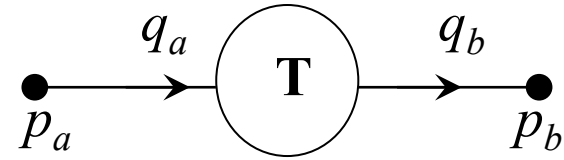
Solve nonlinear system of equations for k_+ s and k_- s

S. Allam and M. Åbom, Investigation of damping and radiation using full plane wave decomposition in ducts. *Journal of Sound and Vibration* 292 (2006) 519-534. doi:10.1016/j.jsv.2005.08.016

Passive two-port measurement techniques

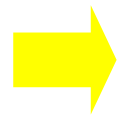


Physical system



Equivalent circuit

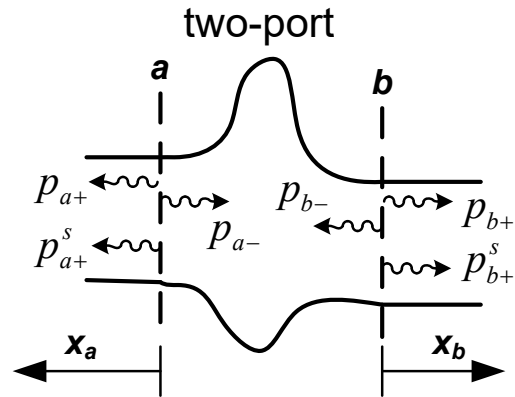
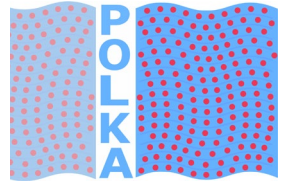
In the frequency domain a (linear) matrix relationship relates the states at a and b . A common choice of state variables is p and q .



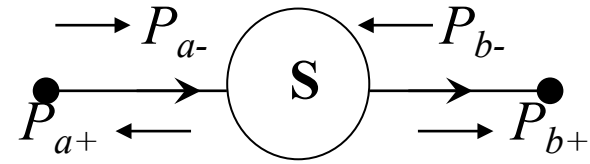
$$\begin{pmatrix} \hat{p}_a \\ \hat{q}_a \end{pmatrix} = \begin{pmatrix} T_{11} & T_{12} \\ T_{21} & T_{22} \end{pmatrix} \begin{pmatrix} \hat{p}_b \\ \hat{q}_b \end{pmatrix}$$

Mathematical model

"Four pole"

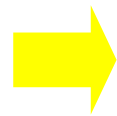


Physical system



Equivalent circuit

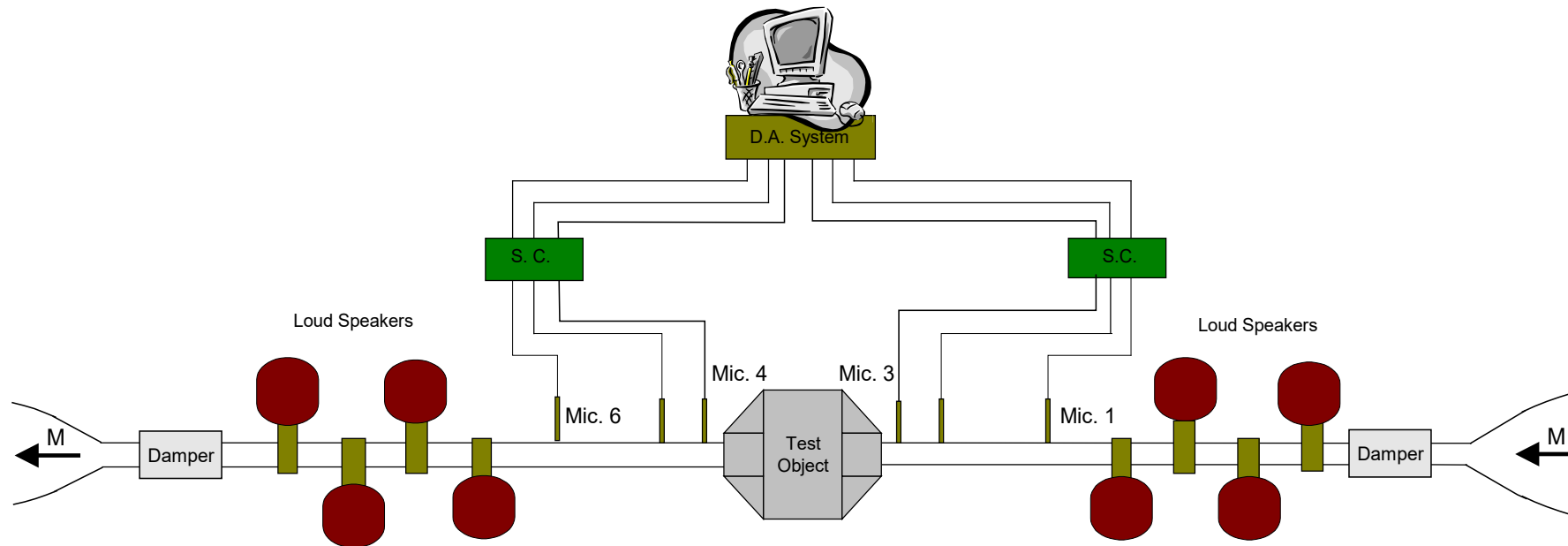
In the frequency domain a (linear) matrix relationship relates the states at a and b . Another common choice of state variables is p_+ and p_- .



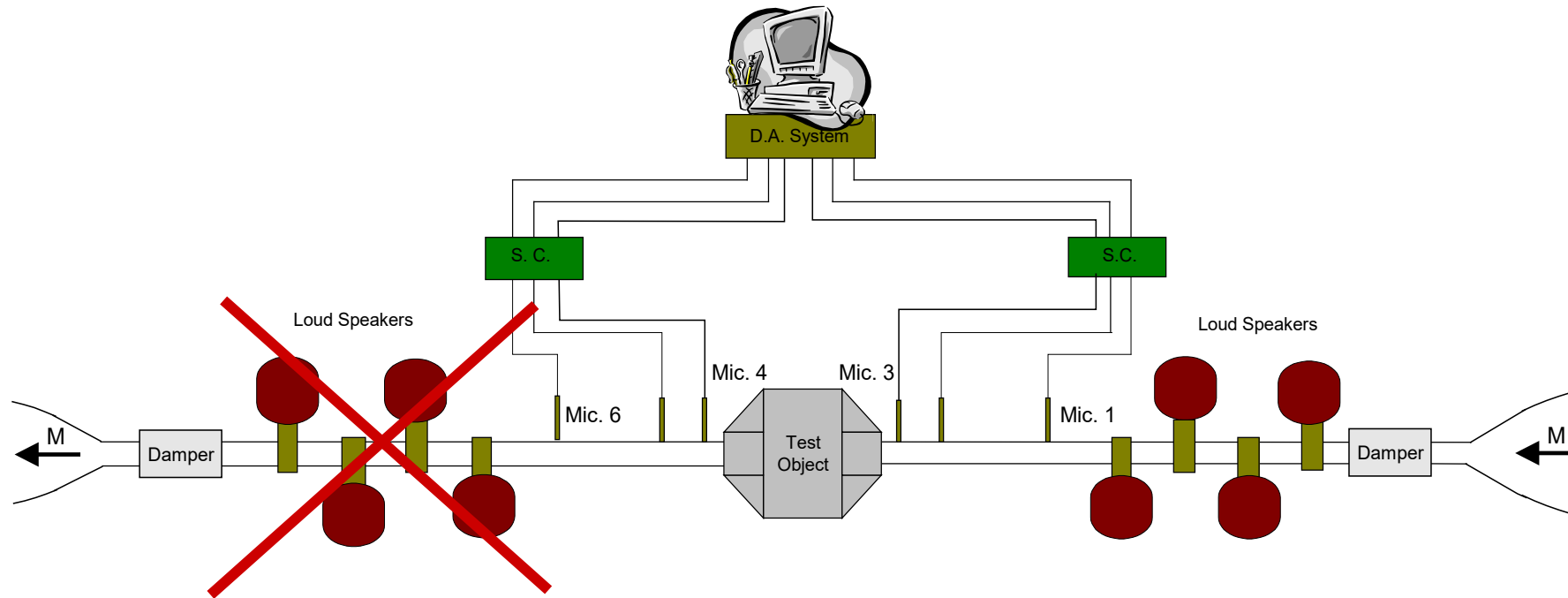
$$\begin{pmatrix} p_{a+} \\ p_{b+} \end{pmatrix} = \begin{pmatrix} S_{11} & S_{12} \\ S_{21} & S_{22} \end{pmatrix} \begin{pmatrix} p_{a-} \\ p_{b-} \end{pmatrix}$$

Mathematical model

"Scattering matrix"



Experimental determination of Two-port data using the Two-source technique

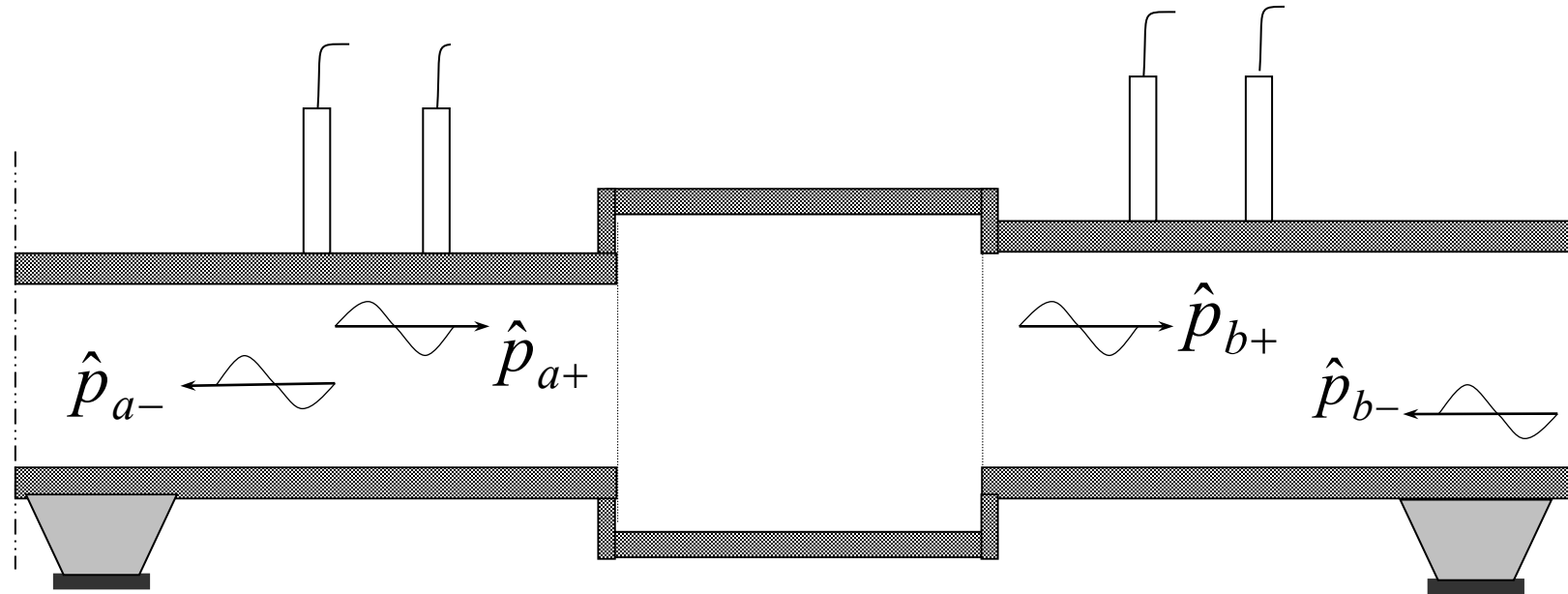


Change acoustic load instead

Experimental determination of Two-port data using the
Two-load technique

Theoretical background

Two-port measurements using the two source technique

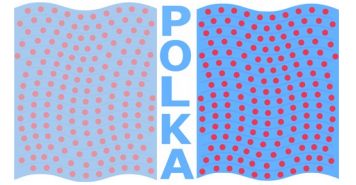


Loudspeaker A

Test object

Loudspeaker B

$$\begin{pmatrix} p_a \\ q_a \end{pmatrix} = \begin{pmatrix} T_{aa} & T_{ab} \\ T_{ba} & T_{bb} \end{pmatrix} \begin{pmatrix} p_b \\ q_b \end{pmatrix} \quad \begin{aligned} p_a &= p_+ \exp(-i k_+^a L_a) + p_- \exp(i k_-^a L_a) \\ q_a &= \frac{A_a}{\rho c} \{ (p_+ \exp(-i k_+^a L_a) - p_- \exp(i k_-^a L_a)) \} \end{aligned}$$



$$\begin{pmatrix} p_a^1 & p_a^2 \\ q_a^1 & q_a^2 \end{pmatrix} = \begin{pmatrix} T_{aa} & T_{ab} \\ T_{ba} & T_{bb} \end{pmatrix} \begin{pmatrix} p_b^1 & p_b^2 \\ q_b^1 & q_b^2 \end{pmatrix}$$

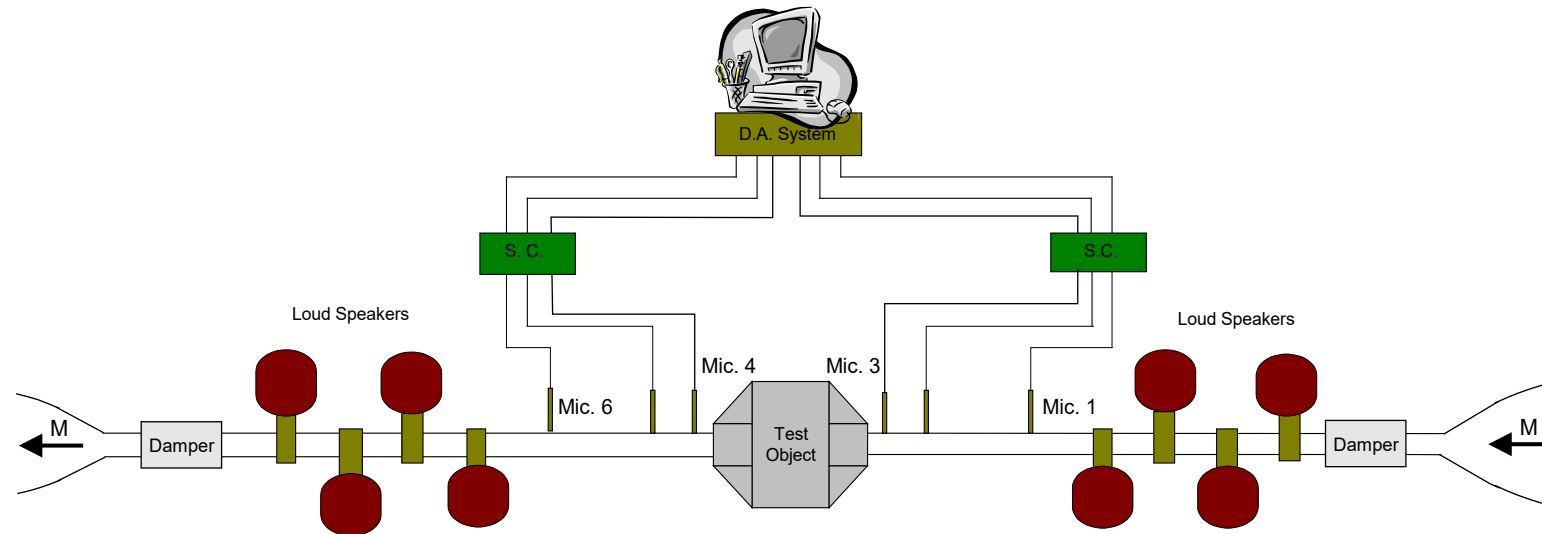
and the two-port matrix is determined from:

$$\begin{pmatrix} T_{aa} & T_{ab} \\ T_{ba} & T_{bb} \end{pmatrix} = \begin{pmatrix} p_a^1 & p_a^2 \\ q_a^1 & q_a^2 \end{pmatrix} \begin{pmatrix} p_b^1 & p_b^2 \\ q_b^1 & q_b^2 \end{pmatrix}^{-1}$$

if

$$\det \begin{pmatrix} p_b^1 & p_b^2 \\ q_b^1 & q_b^2 \end{pmatrix} \neq 0$$

Typical test setup at MWL:

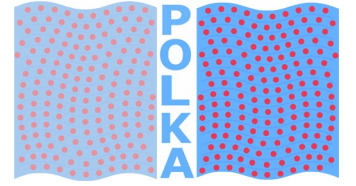


6-8 loudspeakers, two-source technique used,
input signal available as reference

6-12 B&K ¼-inch microphones

Flow speed measured using Pitot tube and hotwire
anemometer. Measurements made before and after the acoustic
measurements.

Flow Noise Supression



Obtain initial good signal-to-noise ratio by:

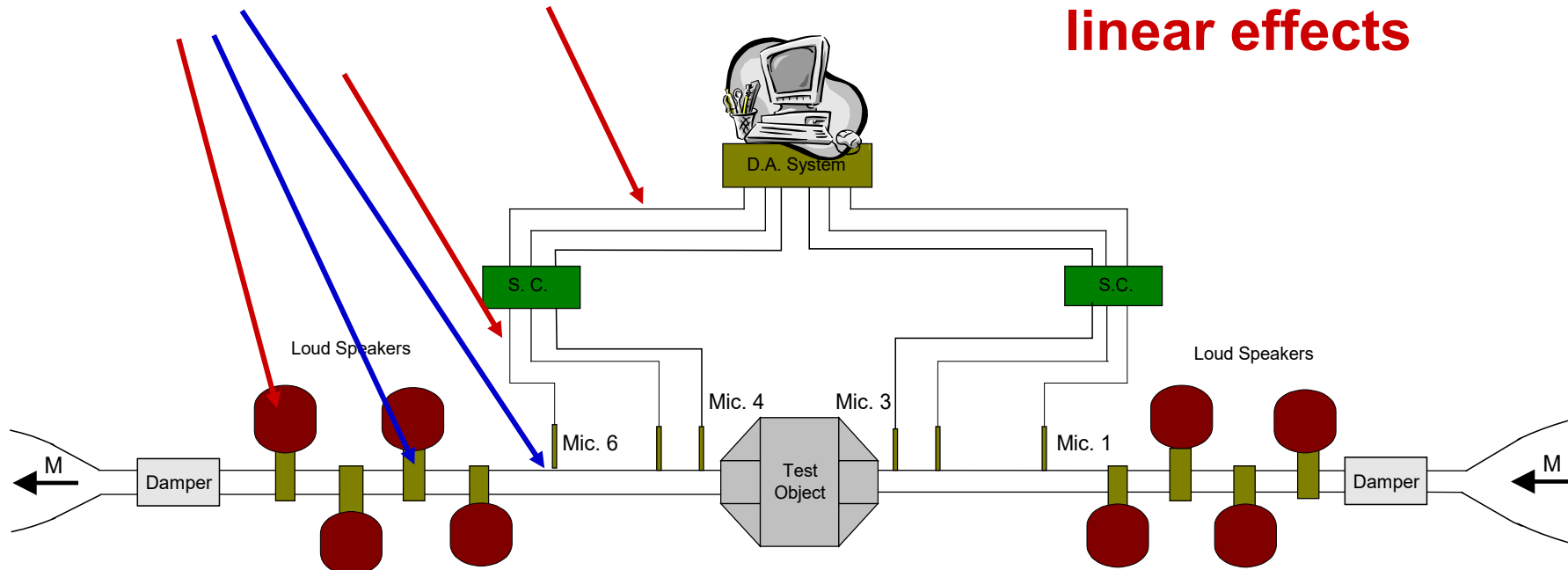
- Increasing level of input signal
- Using multiple loudspeakers
- Using high level loudspeakers
- Concentrating signal energy to narrow frequency bands

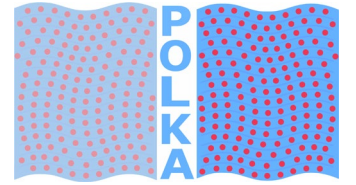
Flow Noise Suppression

Use reference signal and "correlation" techniques:

Possible reference signal locations

Risk for non-linear effects





Signal enhancement techniques

- 1) Frequency domain averaging (**FDA**). Welch's technique.
- 2) Synchronised time domain averaging (**STDA**). Requires deterministic signal + reference (trig) signal.
- 3) Cross-spectrum based frequency domain averaging (**CSFDA**). Requires noise free reference.

Auto-spectrum estimation

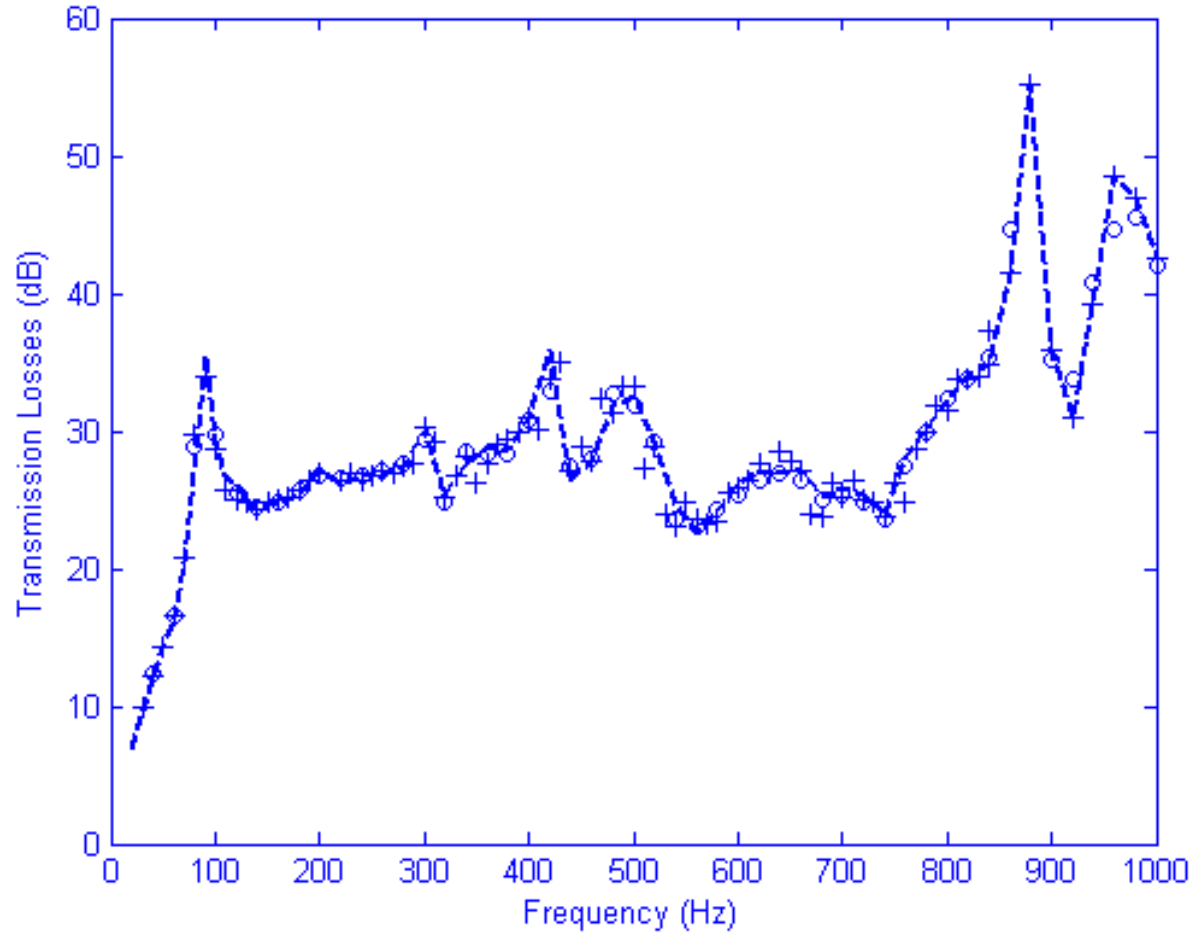
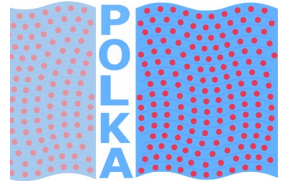
$$\hat{G}_{xx}(\omega) = \frac{\hat{G}_{rx}^*(\omega) \hat{G}_{rx}(\omega)}{\hat{G}_{rr}(\omega)}$$

Averaging

$$\hat{G}_{\gamma x}(\omega) = \frac{1}{N} \sum_{i=1}^N G_{\gamma x}^i(\omega)$$

An improvement in SNR by a factor of N can be expected or $10 \cdot \text{Log}(N)$ dB

Commercial muffler test result



Transmission loss at
 $M=0.26$.

----, random excitation
10000 averages (CSFDA)

++++, sawtooth excitation
10000 averages (STDA)

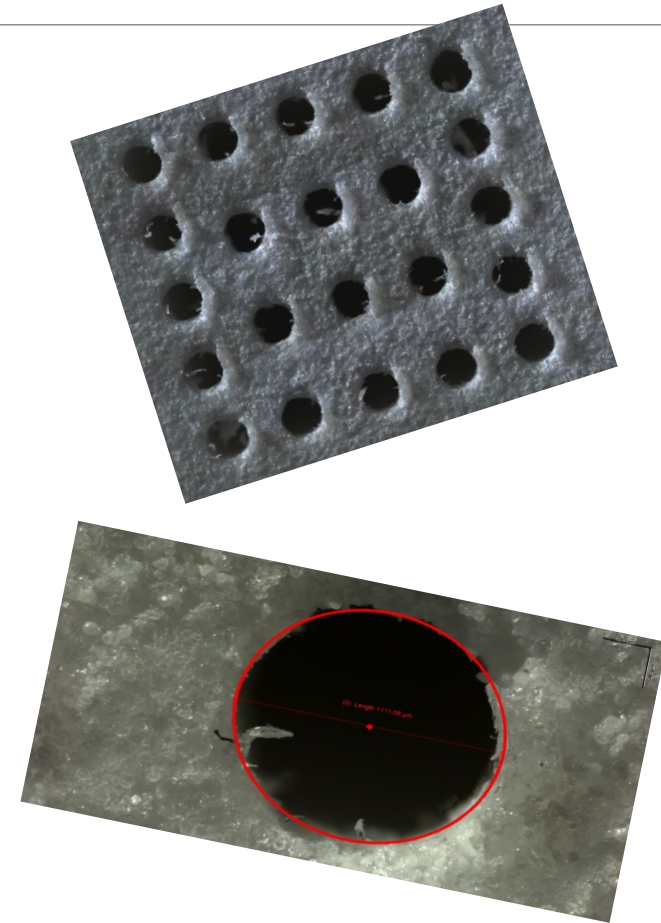
oooo, stepped sine
excitation 400 averages
(CSFDA)



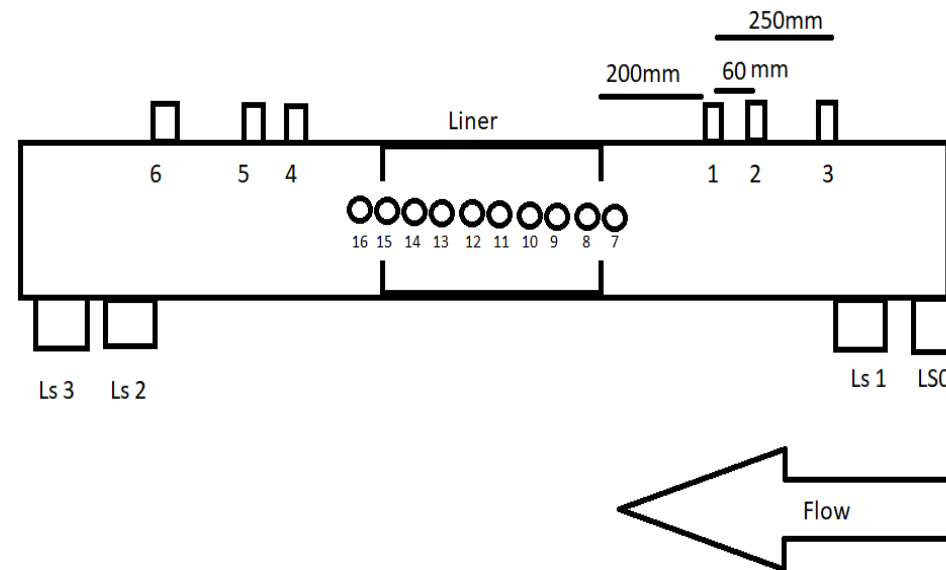
3D-printed liner test

Liners were fabricated using selective layer sintering (SLS) of a PA 2200 polyamide powder on a Formiga P 110 commercial SLS printer with a specified accuracy of 0.2mm \pm 0.002mm/mm. The quality of the printed parts was assessed on an Olympus BX53M microscope.

The back side of the cavities was closed with 20mm thick plywood plates and sealed with silicon.

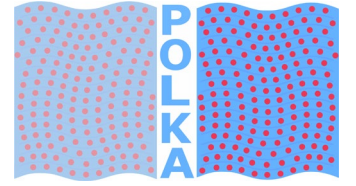


Experimental setup



The microphone distances (in m) were starting from microphone 3: 0, 0.1950, 0.2500, 0.4000, 0.4550, 0.5100, 0.5650, 0.6200, 0.6750, 0.7300, 0.7850, 0.8400, 0.8950, 1.1000, 1.1550, 1.3500.

The distances between microphones 1 and the liner and microphone 4 and the liner was 200 mm.



Experimental analysis techniques

Plane wave scattering matrix

$$\begin{pmatrix} p_{ur} \\ p_{dr} \end{pmatrix} = \begin{bmatrix} \rho_d & \tau_u \\ \tau_d & \rho_u \end{bmatrix} \begin{pmatrix} p_{ui} \\ p_{di} \end{pmatrix}$$

Based on the scattering matrix transmission loss (T_L) and reflection (R), transmission (T) and absorption (A) factors can also be determined:

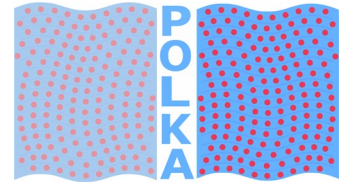
$$T_{Ld} = -10 \text{Log}(\tau_d)$$

$$T_{Lu} = -10 \text{Log}(\tau_u)$$

$$R_d = \frac{(1 - M)^2}{(1 + M)^2} |\rho_d|^2$$

$$R_u = \frac{(1 - M)^2}{(1 + M)^2} |\rho_u|^2$$

Experimental analysis techniques



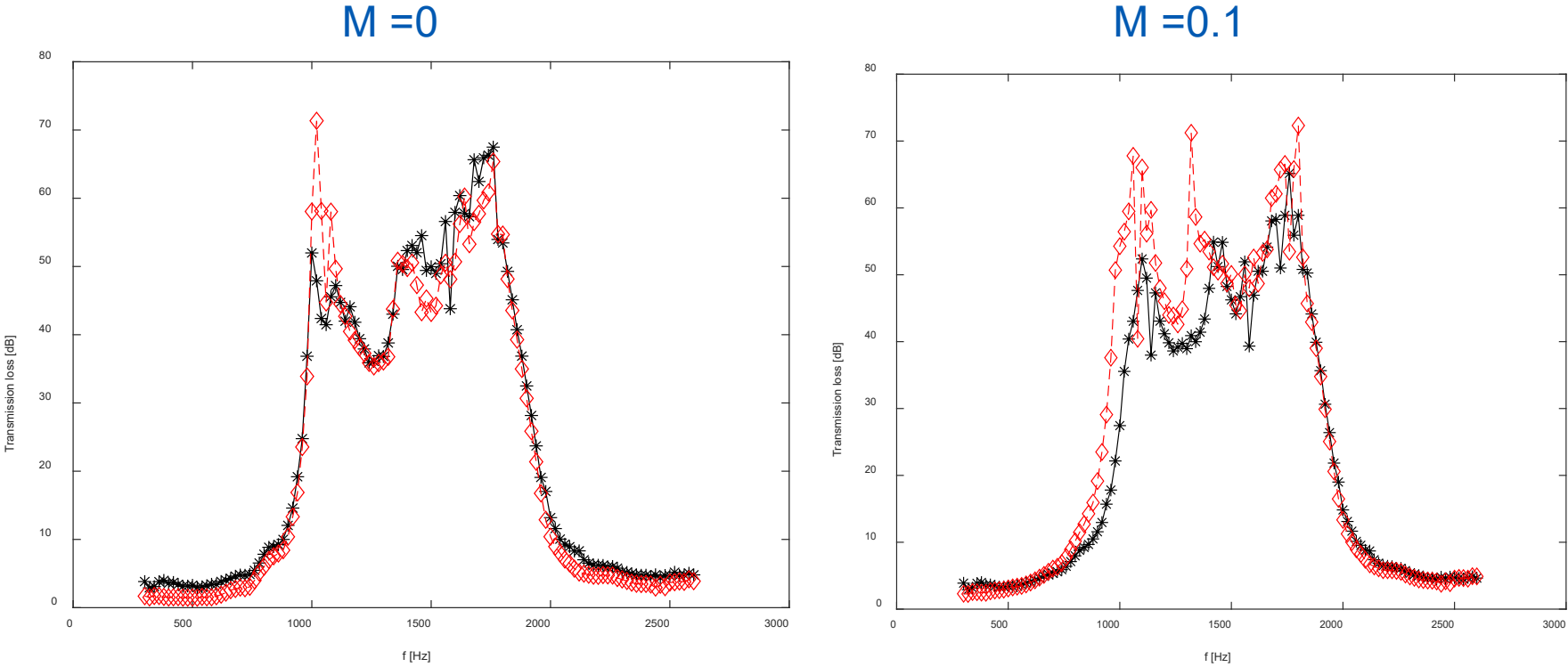
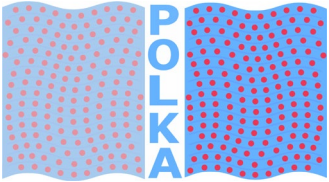
$$T_d = |\tau_d|^2$$

$$T_u = |\tau_u|^2$$

$$A_d = 1 - R_d - T_d$$

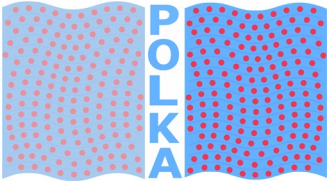
$$A_u = 1 - R_u - T_u$$

Results and discussion – Transmission loss Uniform liner

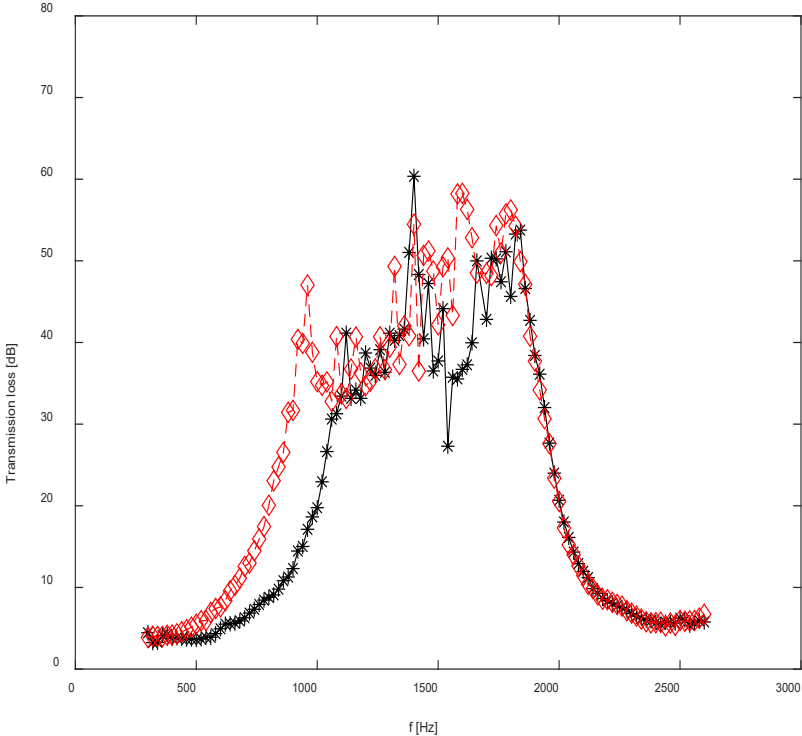


Black – downstream, red - upstream

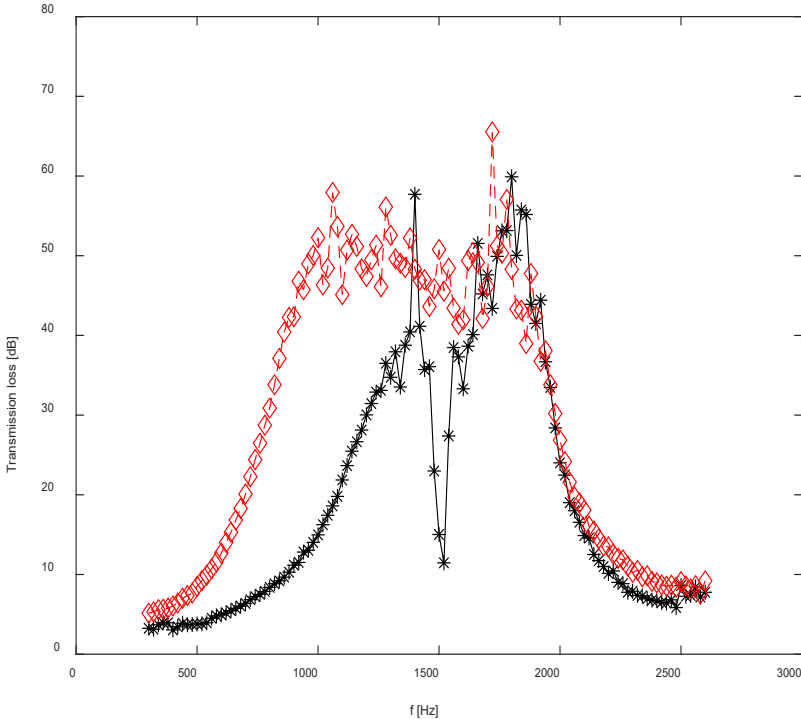
Results and discussion – Transmission loss Uniform liner



M = 0.2

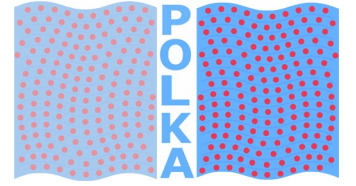


M = 0.3



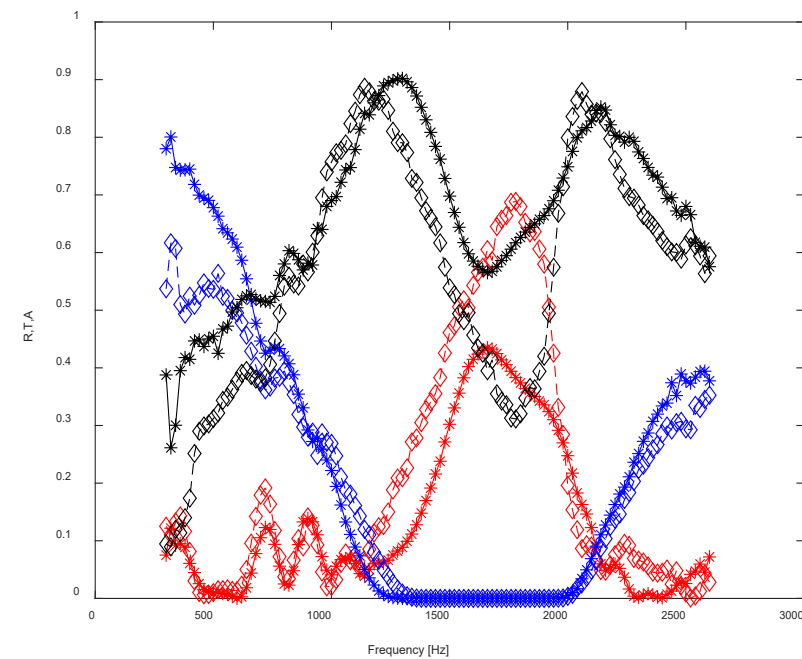
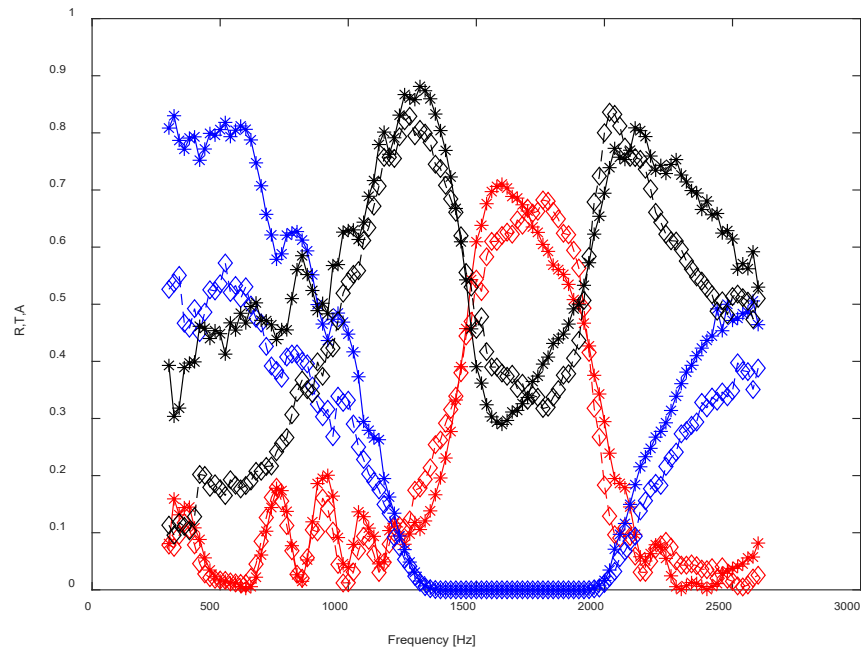
Black – downstream, red - upstream

Results and discussion – Reflection, Transmission, Absorption Uniform liner



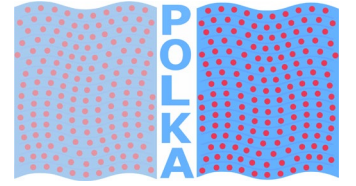
$M = 0$

$M = 0.1$



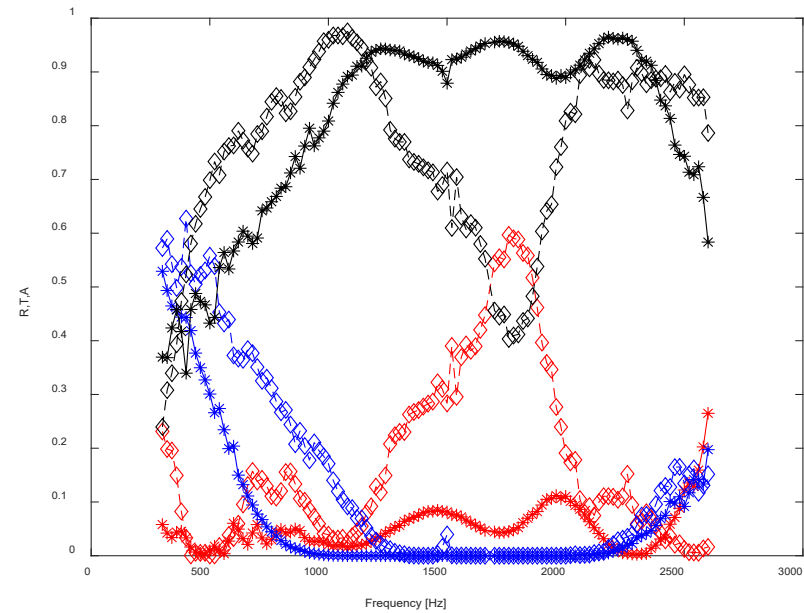
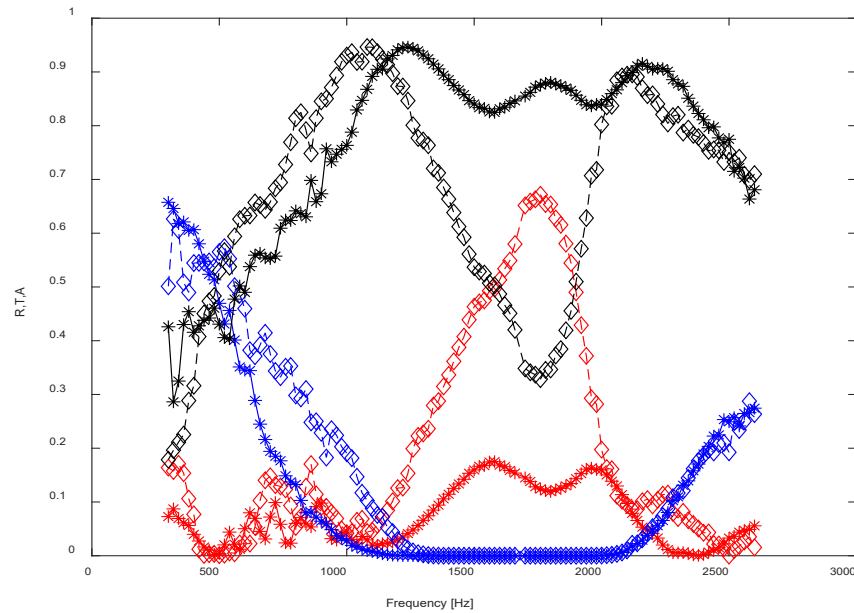
Black – absorption, red – reflection, blue – transmission
Stars – downstream, diamonds – upstream

Results and discussion – Reflection, Transmission, Absorption Uniform liner



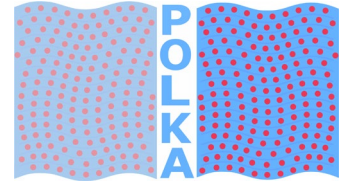
M = 0.2

M = 0.3



Black – absorption, red - reflection, blue - transmission
Stars – downstream, diamonds - upstream

SUMMARY



We have discussed the following topics:

- Short introduction to plane wave duct propagation
- Wave decomposition using the two-microphone method
- Sources of error
- Determination of passive one-port properties
- Determination of passive two-port properties
- Flow noise suppression
- Use of over-determination
- Application examples